

Kepler problem Summary

$$\underline{R} = \text{constant}$$

[if $\underline{R} = 0$, centre of mass is at origin of coordinate system]

$$\underline{r}_1 = \frac{-m_2 \underline{r}}{m_1 + m_2} + \underline{R}$$

$$\underline{r}_2 = \frac{m_1 \underline{r}}{m_1 + m_2} + \underline{R}$$

$$\underline{r} = \frac{a(1-\epsilon^2)}{1+\epsilon\cos\theta} \hat{r}$$

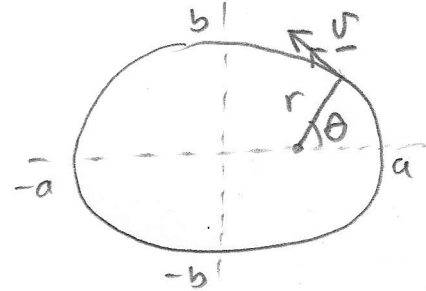
$$r = \frac{a(1-\epsilon^2)}{1+\epsilon\cos\theta}$$

[Orbits are ellipses if $\epsilon < 1$ - Kepler I]

$$b = a\sqrt{1-\epsilon^2}$$

$$\epsilon = \sqrt{1 - \frac{b^2}{a^2}}$$

↑ "Eccentricity"

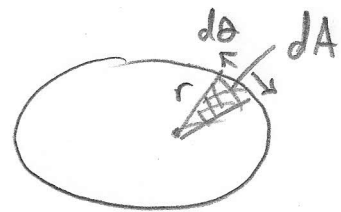


$$\underline{v} = \sqrt{\frac{G(m_1+m_2)}{a(1-\epsilon^2)}} (1+\epsilon\cos\theta) \left(\frac{\epsilon\sin\theta}{1+\epsilon\cos\theta} \hat{r} + \hat{\theta} \right)$$

$$\dot{\theta} = \sqrt{\frac{G(m_1+m_2)}{a^3(1-\epsilon^2)^3}} (1+\epsilon\cos\theta)^2 \quad \therefore \quad t = \int_0^\theta \frac{\sqrt{a^3(1-\epsilon^2)^3}}{\sqrt{G(m_1+m_2)}} \frac{d\theta}{(1+\epsilon\cos\theta)^2}$$

$$E = \frac{-Gm_1m_2}{2a} \quad p^2 = \frac{4\bar{v}^2}{G(m_1+m_2)} a^3 \quad [\text{Kepler III}]$$

$$\frac{dA}{dt} = \frac{1}{2} \sqrt{G(m_1+m_2)(1-\epsilon^2)} a \quad [\text{Kepler II}]$$

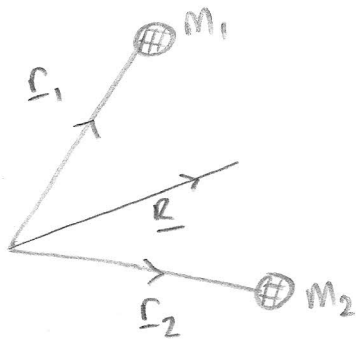


$$J^2 = \frac{Gm_1^2m_2^2(1-\epsilon^2)a}{m_1+m_2}$$

↑

J is the magnitude of the angular momentum. This is a constant of the motion

Two Body KEPLER Problem



Newton's Law of Gravity:

$$m_1 \ddot{\underline{r}}_1 = \frac{GM_1 M_2}{|\underline{r}_2 - \underline{r}_1|^3} (\underline{r}_2 - \underline{r}_1) \quad (1)$$

$$m_2 \ddot{\underline{r}}_2 = -\frac{GM_1 M_2}{|\underline{r}_2 - \underline{r}_1|^3} (\underline{r}_2 - \underline{r}_1) \quad (2)$$

Define

$$\underline{r} = \underline{r}_2 - \underline{r}_1 \quad \therefore \underline{r}_2 = \underline{r} + \underline{r}_1 \quad \text{or} \quad \underline{r}_1 = \underline{r}_2 - \underline{r}$$

$$\underline{R} = \frac{m_1 \underline{r}_1 + m_2 \underline{r}_2}{m_1 + m_2}$$

Also define

$$r = |\underline{r}|$$

← centre of mass of system

so

$$\underline{R} = \frac{m_1 \underline{r}_1 + m_2 \underline{r} + m_2 \underline{r}_1}{m_1 + m_2}$$

$$\underline{R} = \underline{r}_1 + \frac{m_2 \underline{r}}{m_1 + m_2} \quad \therefore$$

$$\underline{r}_1 = \underline{R} - \frac{m_2}{m_1 + m_2} \underline{r} \quad (3)$$

Also

$$\underline{R} = \frac{m_1 \underline{r}_2 - m_1 \underline{r} + m_2 \underline{r}_2}{m_1 + m_2}$$

$$\underline{R} = \underline{r}_2 - \frac{m_1 \underline{r}}{m_1 + m_2} \quad \therefore$$

$$\underline{r}_2 = \underline{R} + \frac{m_1 \underline{r}}{m_1 + m_2} \quad (4)$$

(2) - (1):

$$\ddot{\underline{r}} = \ddot{\underline{r}}_2 - \ddot{\underline{r}}_1 = -\frac{G}{r^3} (m_2 + m_1) \underline{r}$$

ie

$$\ddot{\underline{r}} = -\frac{GM}{r^3} \underline{r}$$

$$M = m_1 + m_2$$

(5)

ie same problem as a small particle orbiting a large mass \$M\$, in other words a one body problem.

Now
$$\underline{R} = \frac{m_1 \underline{r}_1 + m_2 \underline{r}_2}{M}$$

$$\underline{\dot{R}} = \frac{m_1}{M} \underline{\dot{r}}_1 + \frac{m_2}{M} \underline{\dot{r}}_2$$

using (1), (2):
$$m_1 \underline{\ddot{r}}_1 + m_2 \underline{\ddot{r}}_2 = \underline{0}$$

So
$$\underline{\ddot{R}} = 0 \Rightarrow \underline{\dot{R}} = \text{constant} \quad (6)$$

i.e. centre of mass of system moves at a constant velocity.

Let us choose a frame of reference such that $\underline{\dot{R}} = \underline{0}$. This is an inertial frame, so the dynamics of m_1 and m_2 will not be affected.

The angular momentum of the system is

$$\underline{J} = m_1 \underline{r}_1 \times \underline{\dot{r}}_1 + m_2 \underline{r}_2 \times \underline{\dot{r}}_2$$

using $\underline{r}_1 = \underline{R} - \frac{m_2 \underline{r}}{M}$ and $\underline{r}_2 = \underline{R} + \frac{m_1 \underline{r}}{M}$

$$\underline{J} = m_1 \left(\underline{R} - \frac{m_2 \underline{r}}{M} \right) \times \left(-\frac{m_2}{M} \underline{\dot{r}} \right) + m_2 \left(\underline{R} + \frac{m_1 \underline{r}}{M} \right) \times \frac{m_1}{M} \underline{\dot{r}}$$

(since $\underline{\dot{R}} = 0$)

$$\underline{J} = \frac{m_1 m_2}{m_1 + m_2} \underline{r} \times \underline{\dot{r}} \quad (7)$$

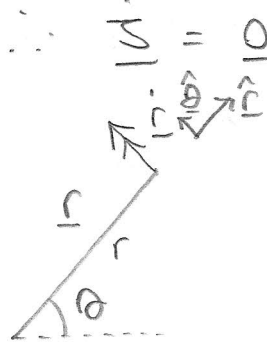
$$\underline{\dot{J}} = \frac{m_1 m_2}{m_1 + m_2} \left(\underline{\dot{r}} \times \underline{\dot{r}} + \underline{r} \times \underline{\ddot{r}} \right) = \underline{0}$$

Since (5):
$$\underline{\ddot{r}} = -\frac{GM}{r^3} \underline{r} \quad \text{and} \quad \underline{\dot{r}} \times \underline{\dot{r}} = 0$$

$$\underline{r} \times \underline{r} = 0$$

(2)

$\dot{\underline{J}} = \underline{0}$ implies \underline{J} is a constant. Define $J = |\underline{J}|$
 \Rightarrow which means motion is planar



Now $\underline{r} = r \hat{r}$ in plane polars
 $\dot{\underline{r}} = \dot{r} \hat{r} + r \dot{\theta} \hat{\theta}$
 $\therefore |\dot{\underline{r}}|^2 = \dot{r}^2 + r^2 \dot{\theta}^2$ { $|\dot{\underline{r}}|^2 = \dot{\underline{r}} \cdot \dot{\underline{r}}$ }

For our system, the total energy is

$$E = \frac{1}{2} M_1 |\dot{\underline{r}}_1|^2 + \frac{1}{2} M_2 |\dot{\underline{r}}_2|^2 - \frac{GM_1 M_2}{r}$$

$$\dot{\underline{r}}_1 = -\frac{M_2 \dot{\underline{r}}}{M} \quad \text{and} \quad \dot{\underline{r}}_2 = \frac{M_1 \dot{\underline{r}}}{M}$$

$$\left[\begin{aligned} &M_1 M_2^2 + M_2 M_1^2 \\ &= M_1 M_2 (M_1 + M_2) \\ &= M_1 M_2 M \end{aligned} \right]$$

$$\therefore E = \frac{1}{2} \frac{M_1 M_2^2}{M^2} |\dot{\underline{r}}|^2 + \frac{1}{2} \frac{M_2 M_1^2}{M^2} |\dot{\underline{r}}|^2 - \frac{GM_1 M_2}{r}$$

$$E = \frac{1}{2} \frac{M_1 M_2}{M} |\dot{\underline{r}}|^2 - \frac{GM_1 M_2}{r}$$

$$E = \frac{M_1 M_2}{2M} (\dot{r}^2 + r^2 \dot{\theta}^2) - \frac{GM_1 M_2}{r} \quad (P)$$

Now $\underline{J} = \frac{M_1 M_2}{M} r \hat{r} \times (\dot{r} \hat{r} + r \dot{\theta} \hat{\theta})$

$$\therefore \underline{J} = \frac{M_1 M_2}{M} r^2 \dot{\theta} (\hat{r} \times \hat{\theta})$$

$$\therefore J^2 = \frac{M_1^2 M_2^2}{M^2} r^4 \dot{\theta}^2$$

$$\therefore \dot{\theta}^2 = \frac{M^2 J^2}{M_1^2 M_2^2 r^4} \quad (P^1/2)$$

$$\therefore E = \frac{M_1 M_2}{2M} \left(\dot{r}^2 + \frac{M^2 J^2}{M_1^2 M_2^2 r^2} \right) - \frac{GM_1 M_2}{r}$$

(3)

Now, for reasons that will become clear later, consider a variable change

$$\boxed{u = \frac{1}{r}}$$

$$\dot{u} = -\frac{1}{r^2} \dot{r} \Rightarrow \dot{r} = -r^2 \dot{u} = -\frac{\dot{u}}{u^2}$$

$$\therefore E = \frac{m_1 m_2}{2M} \left(\frac{\dot{u}^2}{u^4} + \left(\frac{MS}{m_1 m_2} \right)^2 u^2 \right) - GM_1 m_2 u$$

Now we want the trajectory of the masses, so $u = u(\theta)$ is desired. (Rather than $u(t)$, and $\theta(t)$).

$$\text{Now } \frac{du}{d\theta} = \frac{du}{dt} \times \frac{dt}{d\theta} = \frac{\dot{u}}{\dot{\theta}}$$

$$\therefore \dot{u}^2 = \dot{\theta}^2 \left(\frac{du}{d\theta} \right)^2 \quad \leftarrow \text{Eqn (8)}$$

$$\text{Using } \dot{\theta}^2 = \left(\frac{MS}{m_1 m_2 r^2} \right)^2 = \left(\frac{MS}{m_1 m_2} \right)^2 u^4$$

$$\Rightarrow \dot{u}^2 = \left(\frac{du}{d\theta} \right)^2 \left(\frac{MS}{m_1 m_2} \right)^2 u^4$$

$$\therefore E = \frac{M}{2m_1 m_2} S^2 \left(\left(\frac{du}{d\theta} \right)^2 + u^2 \right) - GM_1 m_2 u$$

$$\text{Now conservation of energy } \Rightarrow \frac{dE}{dt} = \frac{dE}{d\theta} = 0$$

$$\therefore 0 = \frac{M}{2m_1 m_2} S^2 \left(2 \frac{du}{d\theta} \frac{d^2 u}{d\theta^2} + 2u \frac{du}{d\theta} \right) - GM_1 m_2 \frac{du}{d\theta}$$

$$\text{Since } \frac{du}{d\theta} \neq 0 \quad \forall \theta$$

$$\Rightarrow \frac{M}{m_1 m_2} S^2 \left(\frac{d^2 u}{d\theta^2} + u \right) = GM_1 m_2$$

$$\boxed{\frac{d^2 u}{d\theta^2} + u = \frac{GM_1^2 m_2^2}{(m_1 + m_2) S^2}} \quad (9)$$

Now, an ellipse has polar equation

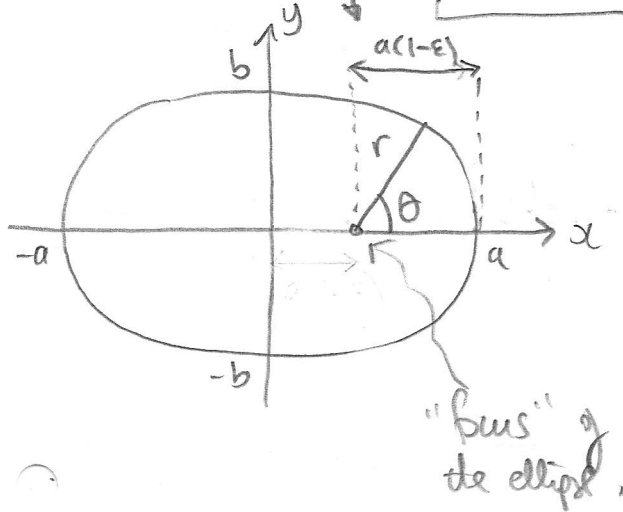
when $\theta=0$, $r = \frac{a(1+\epsilon)(1-\epsilon)}{1+\epsilon}$
 $= a(1-\epsilon)$

$$r = \frac{a(1-\epsilon^2)}{1+\epsilon\cos\theta}$$

a is the "Semi-major axis"

ϵ is the eccentricity

$$\epsilon = \sqrt{1 - \frac{b^2}{a^2}}$$



Note

$$b = a\sqrt{1-\epsilon^2}$$

↑
 i.e. circular orbit
 $\Rightarrow \epsilon=0$
 i.e. $a=b$

$$u(\theta) = \frac{1 + \epsilon\cos\theta}{a(1-\epsilon^2)}$$

Substituting into (9)

$$\frac{-\epsilon\cos\theta}{a(1-\epsilon^2)} + \frac{1 + \epsilon\cos\theta}{a(1-\epsilon^2)} = \frac{GM_1M_2}{(m_1+m_2)\mathcal{J}^2}$$

$$\frac{d^2a}{d\theta^2} \Rightarrow \frac{1}{a(1-\epsilon^2)} = \frac{GM_1M_2}{(m_1+m_2)\mathcal{J}^2}$$

$$\mathcal{J}^2 = \frac{GM_1M_2(1-\epsilon^2)a}{m_1+m_2} \quad (10)$$

i.e. (9) accepts ellipses as solutions **THUS PROVING KEPLER'S FIRST LAW**, that is

"Bound orbits of a two-mass system driven by an inverse square law of gravitational attraction are ellipses".

Now
$$E = \frac{M}{2m_1 m_2} J^2 \left(\left(\frac{-\sin\theta \epsilon}{a(1-\epsilon^2)} \right)^2 + \left(\frac{1+\epsilon\cos\theta}{a(1-\epsilon^2)} \right)^2 \right) + \dots$$

$\left(\frac{du}{d\theta} \right)^2 \dots - \frac{GM_1 M_2 (1+\epsilon\cos\theta)}{a(1-\epsilon^2)}$

$$E = \frac{M}{2m_1 m_2} \times \frac{GM_1^2 M_2^2 (1-\epsilon^2) a}{(m_1+m_2)} \left\{ \frac{\epsilon^2 \sin^2\theta + 1 + \epsilon^2 \cos^2\theta + 2\epsilon\cos\theta}{a^2(1-\epsilon^2)^2} \right\}$$

$[M = m_1 + m_2]$

$$- \frac{GM_1 M_2 (1+\epsilon\cos\theta)}{a(1-\epsilon^2)}$$

$$\Rightarrow E = \frac{GM_1 M_2}{2a} \left\{ \frac{\epsilon^2}{1-\epsilon^2} + \frac{1+2\epsilon\cos\theta}{1-\epsilon^2} \right\} - \frac{GM_1 M_2 (1+\epsilon\cos\theta)}{a(1-\epsilon^2)}$$

$$= \frac{GM_1 M_2}{2a} \left\{ \frac{\epsilon^2 + 1 + 2\epsilon\cos\theta - 2 - 2\epsilon\cos\theta}{1-\epsilon^2} \right\}$$

$E = - \frac{GM_1 M_2}{2a}$

(11)

In Summary so far:

$$r(\theta) = \frac{a(1-\epsilon^2)}{1+\epsilon\cos\theta}$$

$$E = - \frac{GM_1 M_2}{2a}$$

$$\underline{r} = r(\theta) \hat{r}$$

$$J^2 = \frac{GM_1^2 M_2^2 (1-\epsilon^2) a}{m_1 + m_2}$$

$$\underline{r} = r\cos\theta \hat{x} + r\sin\theta \hat{y}$$

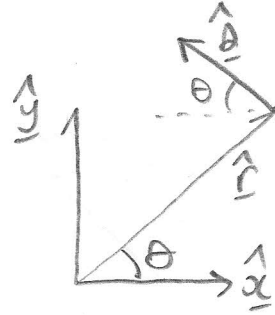
$$\underline{r}_1 = - \frac{m_2}{m_1 + m_2} \underline{r}$$

⑥ $\dot{\theta}^2 = \frac{(m_1 + m_2)^2 J^2}{m_1^2 m_2^2 r^4}$

$$\underline{r}_2 = \frac{m_1}{m_1 + m_2} \underline{r}$$

$$\underline{v} = \dot{r} \hat{r} + r \dot{\theta} \hat{\theta} = \dot{r}$$

$$\hat{\theta} = -\sin\theta \hat{x} + \cos\theta \hat{y}$$



$$\dot{r} = -\frac{a(1-\epsilon^2)}{(1+\epsilon\cos\theta)^2} (-\epsilon\sin\theta) \dot{\theta} \hat{r} + \frac{a(1-\epsilon^2)}{1+\epsilon\cos\theta} \dot{\theta} \hat{\theta}$$

$$\underline{v} = \frac{a(1-\epsilon^2) \dot{\theta}}{1+\epsilon\cos\theta} \left(\frac{\epsilon\sin\theta}{1+\epsilon\cos\theta} \hat{r} + \hat{\theta} \right)$$

$$\left[\begin{array}{l} \sin\theta \\ r = \frac{a(1-\epsilon^2)}{1+\epsilon\cos\theta} \end{array} \right]$$

Now

$$\dot{\theta}^2 = \frac{(m_1+m_2)^2}{m_1^2 m_2^2 r^4} \frac{G m_1^2 m_2^2 (1-\epsilon^2) a}{m_1+m_2}$$

$$\Rightarrow \dot{\theta}^2 = G(m_1+m_2)(1-\epsilon^2)a \times$$

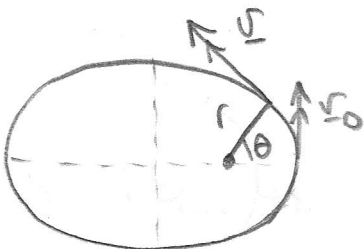
$$\frac{(1+\epsilon\cos\theta)^4}{a^4(1-\epsilon^2)^4}$$

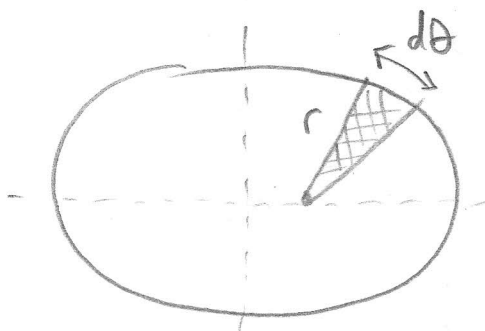
$$\dot{\theta} = \sqrt{\frac{G(m_1+m_2)}{a^3(1-\epsilon^2)^3}} (1+\epsilon\cos\theta)^2 \quad (12)$$

$$\underline{v} = \sqrt{\frac{G(m_1+m_2)}{a(1-\epsilon^2)}} (1+\epsilon\cos\theta) \left(\frac{\epsilon\sin\theta}{1+\epsilon\cos\theta} \hat{r} + \hat{\theta} \right) \quad (13)$$

Note when $\theta=0$

$$\underline{v}_{\theta=0} = \sqrt{\frac{G(m_1+m_2)}{a(1-\epsilon^2)}} (1+\epsilon) \hat{\theta}$$





Area swept out by orbit occurs at rate s.t $dA = \frac{1}{2} r^2 d\theta$

$$\therefore \frac{dA}{dt} = \frac{1}{2} r^2 \dot{\theta}$$

$$\begin{aligned} \text{Now } r^2 \dot{\theta} &= \frac{a^2(1-\epsilon^2)}{(1+\epsilon \cos \theta)^2} \sqrt{\frac{G(m_1+m_2)}{a^3(1-\epsilon^2)^3}} (1+\epsilon \cos \theta)^2 \\ &= \sqrt{G(m_1+m_2)a(1-\epsilon^2)} \end{aligned}$$

$$\text{So } \boxed{\frac{dA}{dt} = \frac{1}{2} \sqrt{G(m_1+m_2)(1-\epsilon^2)a}}$$

KEPLER'S SECOND LAW: "The rate of area $\frac{dA}{dt}$ swept out by the elliptical orbit from the focus of the ellipse is a constant"

Now since $\frac{dA}{dt}$ is a constant, the orbital period

$$P = \frac{\pi ab}{\frac{dA}{dt}}$$

$$\text{Now } b = a\sqrt{1-\epsilon^2}$$

$$\therefore P = \frac{\pi a^2 \sqrt{1-\epsilon^2}}{\frac{1}{2} \sqrt{G(m_1+m_2)(1-\epsilon^2)a}}$$

$$\Rightarrow \boxed{P^2 = \frac{4\pi^2}{G(m_1+m_2)} a^3}$$

KEPLER'S THIRD LAW

$$P^2 \propto a^3$$